

AA TT PRO 01a

Name of Assessed Person: Registration:

UNIT MEA316: Inspect, Test and Troubleshoot Rotary Wing Rotor and Control Systems and Components												
			No. of Entries	:	L	2	<u>)</u>		3			
			Tail / Job No.									
1.	a.		LAME Sign.									
			Date									
			Simulated	Yes	No	Yes	No	Yes	No			
		. Rotor Heads, Swash Plates, Tail Rotor Pitch Control Assemblies	No. of Entries	:	_	2	2	:	3			
	b.		Tail / Job No.									
Inspect Rotor and Rotor Control Systems and			LAME Sign.									
Components			Date									
Components			Simulated	Yes	No	Yes	No	Yes	No			
			No. of Entries	:	<u>L</u>	2	2	3	3			
			Tail / Job No.									
	C.	Mechanical, Powered Flight Control Components	LAME Sign.									
			Date									
			Simulated	Yes	No	Yes	No	Yes	No			

- 1.1 Isolation and warning signs are fitted / installed to the system or related systems and the aircraft configured for safe system inspection and operation in accordance with relevant aircraft publications/maintenance regulations orders and standards and practices.
- 1.2 **Rotor and rotor control system** is visually or physically checked/inspected for external signs of defects in accordance with relevant aircraft publications maintenance regulations/orders and standards and practices while observing all relevant work health and safety (WHS) requirements.
- 1.3 Defects are identified and recorded in accordance with standard enterprise procedures.



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UNIT MEA316: Inspect, Test a	nd Tro	ubleshoot Rotary Wing Rotor and Control Systems and Components							
		d. Main Rotor, Intermediate or Tail Rotor Gearboxes L	No. of Entries	1	•	2		3	3
1. Cont'd			Tail / Job No.						
	d.		LAME Sign.						
			Date						
Inspect Rotor and Rotor			Simulated	Yes	No	Yes	No	Yes	No
Control Systems and			No. of Entries	1		2		3	3
Components			Tail / Job No.						
	e.	Drive Shafts and Couplings	LAME Sign.						
			Date						
			Simulated	Yes	No	Yes	No	Yes	No

- 1.1 Isolation and warning signs are fitted / installed to the system or related systems and the aircraft configured for safe system inspection and operation in accordance with relevant aircraft publications/maintenance regulations orders and standards and practices.
- 1.2 **Rotor and rotor control system** is visually or physically checked/inspected for external signs of defects in accordance with relevant aircraft publications maintenance regulations/orders and standards and practices while observing all relevant work health and safety (WHS) requirements.
- 1.3 Defects are identified and recorded in accordance with standard enterprise procedures.



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			No. of Entries	1	L	2		(1)	3				
			Tail / Job No.										
2.	a.	Main Rotor Blades, Tail Rotor Blades	LAME Sign.										
			Date										
		Simulated	Simulated	Yes	No	Yes	No	Yes	No				
		Rotor Heads, Swash Plates, Tail Rotor Pitch Control Assemblies	No. of Entries	1	_	2) -	3	3				
			Tail / Job No.										
Ground Test Rotor and Rotor	b.		LAME Sign.										
Control System			Date										
			Simulated	Yes	No	Yes	No	Yes	No				
			No. of Entries	1	_	2		3	3				
			Tail / Job No.										
	C.	Mechanical, Powered Flight Control Components	LAME Sign.										
			Date										
			Simulated	Yes	No	Yes	No	Yes	No				

- 2.1 Aircraft and system prepared in accordance with relevant aircraft publications/maintenance regulations orders and standards and practices, for the operation of engine and rotor system.
- 2.2 Rotor and rotor control system are functionally tested, in accordance with relevant aircraft publications maintenance regulations/orders and standards and practices, for evidence of malfunction.
- 2.3 System calibration or adjustments are performed in accordance with relevant aircraft publications/ maintenance regulations/orders and standards and practices.



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		Main Rotor, Intermediate or Tail Rotor Gearboxes	No. of Entries	1		2		3	3
			Tail / Job No.						
	d.		LAME Sign.						
			Date						
2. Cont'd Ground Test Rotor and Rotor			Simulated	Yes	No	Yes	No	Yes	No
Control System			No. of Entries	1		2		3	3
Control System		Drive Shafts and Couplings	Tail / Job No.						
	e.		LAME Sign.						
			Date						
			Simulated	Yes	No	Yes	No	Yes	No

- 2.1 Aircraft and system prepared in accordance with relevant aircraft publications/maintenance regulations orders and standards and practices, for the operation of engine and rotor system.
- 2.2 Rotor and rotor control system are functionally tested, in accordance with relevant aircraft publications maintenance regulations/orders and standards and practices, for evidence of malfunction.
- 2.3 System calibration or adjustments are performed in accordance with relevant aircraft publications/ maintenance regulations/orders and standards and practices.



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			No. of Entries	1	-	2	2		3
			Tail / Job No.						
	a.	Main Rotor Blades, Tail Rotor Blades	LAME Sign.						
	Date	Date							
			Simulated	Yes	No	Yes	No	Yes	No
		Rotor Heads, Swash Plates, Tail Rotor Pitch Control Assemblies LA Da	No. of Entries	1	-	2	2		3
	b.		Tail / Job No.						
3. Prepare for Troubleshooting			LAME Sign.						
Prepare for froubleshooting			Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries	1	-	2	2		3
			Tail / Job No.						
	c.	c. Mechanical, Powered Flight Control Components	LAME Sign.						
			Date						
		Simulated	Yes	No	Yes	No	Yes	No	

- 3.1 Relevant aircraft publications and modification status, including system defect reports, are interpreted to identify unserviceability.
- ** Note: Troubleshooting: involves the use of fault finding charts or similar, to line replacement level.



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		. Main Rotor, Intermediate or Tail Rotor Gearboxes	No. of Entries	1	•	2		(1)	3
			Tail / Job No.						
	d.		LAME Sign.						
			Date						
3. Cont'd			Simulated	Yes	No	Yes	No	Yes	No
Prepare for Troubleshooting			No. of Entries	1		2		3	3
		Drive Shafts and Couplings	Tail / Job No.						
	e.		LAME Sign.						
			Date						
			Simulated	Yes	No	Yes	No	Yes	No

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			No. of Entries	1	Ĺ	2		***	3			
			Tail / Job No.									
	a.	Main Rotor Blades, Tail Rotor Blades	LAME Sign.									
4.			Date									
	Simulated	Simulated	Yes	No	Yes	No	Yes	No				
			No. of Entries	1	_	2) -		3			
	b.	b. Rotor Heads, Swash Plates, Tail Rotor Pitch Control Assemblies	Tail / Job No.									
Troubleshoot Rotor and Rotor			LAME Sign.									
Control Systems			Date									
			Simulated	Yes	No	Yes	No	Yes	No			
			No. of Entries	1	_	2) -	3	3			
			Tail / Job No.									
	c.	Mechanical, Powered Flight Control Components	LAME Sign.									
			Date									
			Simulated	Yes	No	Yes	No	Yes	No			

- 4.1 Available information from aircraft maintenance documentation, inspection and test results is used to assist in fault determination.
- 4.2 Relevant aircraft publication fault diagnosis guide and logical processes are used to ensure efficient and accurate *troubleshooting* to line replacement level.
- 4.3 Specialist advice is obtained to assist with the troubleshooting process.
- 4.4 Rotor and rotor control system faults are located and the causes of the faults are clearly identified and recorded in aircraft maintenance documentation in accordance with standard enterprise procedures.
- 4.5 Fault rectification requirements are determined.

^{**} Note: Troubleshooting: involves the use of fault finding charts or similar, to line replacement level.



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		. Main Rotor, Intermediate or Tail Rotor Gearboxes . Drive Shafts and Couplings	No. of Entries	1	•	2		(1)	3
			Tail / Job No.						
	d.		LAME Sign.						
			Date						
4. Cont'd Troubleshoot Rotor and Rotor			Simulated	Yes	No	Yes	No	Yes	No
Control Systems			No. of Entries	1		1 2		3	
Control Systems			Tail / Job No.						
	e.		LAME Sign.						
			Date						
			Simulated	Yes	No	Yes	No	Yes	No

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- 4.2 Relevant aircraft publication fault diagnosis guide and logical processes are used to ensure efficient and accurate *troubleshooting* to line replacement level.
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Name of Assessed Person: Registration:

Certification of Underpinning Knowledge and Skills to Inspect, Test and Troubleshoot Rotary Wing Rotor and Control Systems and Components

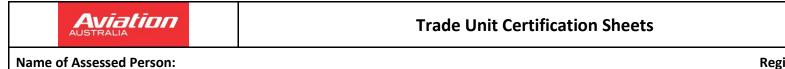
A person cannot be assessed as competent until it can be demonstrated to the satisfaction of the workplace assessor that the relevant elements and performance criteria of the unit of competency are being achieved under routine supervision on each type of system and on at least one (1) component of each group listed in the assessment conditions a) to e) that are applicable to the enterprise. This shall be established via the records in the Log of Industrial Experience and Achievement or, where appropriate, an equivalent Industry Evidence Guide (for details refer to the Companion Volume Assessment Guidelines).

UNIT MEA316: Inspect, Test and Troubleshoot Rotary Wing Rotor and Control Systems and Components	Date/ MTO Stamp
Evidence has been confirmed of the attainment of the following pre-requisite units of competency (as they are related	
to attainment of the elements of competency specified in this unit).	
308	
Evidence has been confirmed of the knowledge requirements for this unit as delivered by a CASR 147 Approved Organisation.	
OR	
Assessment has been conducted to determine that the underpinning knowledge and skills have been achieved in accordance with the Competency Unit.	

Certification of Unit Completion

I certify that at the time of this assessment I have reviewed the candidates' evidence of experiences for the application of skills and knowledge that it meets the requirements specified in the elements and criteria for this unit of competency.

Signed:	Assessor No.	r	MTO: _	Date:	



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Registration:

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