

AA TT PRO 01a

Name of Assessed Person: Registration:

UNIT MEA320: Test and tro	ublesl	noot aircraft hydro-mechanical, gaseous and landing gear systems and	components						
			No. of Entries	1		2	<u>)</u>	(1)	3
	a.	Hydraulic systems - hydraulic accumulators, filters, reservoirs,	Tail / Job No.						
		valves, pumps, motors, actuators, regulators and direct reading	LAME Sign.						
		gauges	Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries	1		2	2	(1)	3
			Tail / Job No.						
	b.	Hydraulic system rigid and flexible pipelines, hoses and fittings	LAME Sign.						
			Date						
1.			Simulated	Yes	No	Yes	No	Yes	No
Prepare for troubleshooting.			No. of Entries	1		2	<u>)</u>	3	3
	c.	Fuel systems - filters, valves, pumps, rigid and flexible storage	Tail / Job No.						
	C.	cells/tanks	LAME Sign.						
		ceris/ turiks	Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries	1		2	<u> </u>		3
			Tail / Job No.						
	d.	Fuel system rigid and flexible pipelines, hoses and fittings	LAME Sign.						
	, , , , , , , , , , , , , , , , , , , ,	Date							
			Simulated	Yes	No	Yes	No	Yes	No

# Performance Criteria:



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UNIT MEA320: Test and tro	ubles	hoot aircraft hydro-mechanical, gaseous and landing gear systems and	components						
			No. of Entries	-	L	12	2	(1)	3
			Tail / Job No.						
	e.	Landing gear systems – retraction systems *(Note 1)	LAME Sign.						
			Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries		L	2	2	3	3
			Tail / Job No.						
	f.	Landing gear systems - steering systems *(Note 1)	LAME Sign.						
			Date						
1. Cont'd			Simulated	Yes	No	Yes	No	Yes	No
Prepare for troubleshooting.			No. of Entries	:	L	2	2	3	3
	_	Landing goar systems, brake systems, including anti-skid, where	Tail / Job No.						
	g.	Landing gear systems - brake systems, including anti-skid, where applicable *(Note 1)	LAME Sign.						
		applicable (Note 1)	Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries		L	2	2	3	3
			Tail / Job No.						
	h.	Landing gear components – wheel assembly *(Note 1)	LAME Sign.						
		[	Date						
			Simulated	Yes	No	Yes	No	Yes	No

### **Performance Criteria:**

1.1 Relevant maintenance documentation and modification status, including system defect/service difficulty reports, where relevant, are interpreted to identify an unserviceability.

\*(Note 1) Not applicable to aircraft fitted with skids or floats



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			No. of Entries		1	2	2	:	3
			Tail / Job No.						
	i.	Landing gear components – brake units *(Note 1)	LAME Sign.						
			Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries		1	2	2	3	3
			Tail / Job No.						
	j.	Landing gear components – struts/oleos *(Note 1)	LAME Sign.						
		Date Simulated	Date						
1. Cont'd			Simulated	Yes	No	Yes	No	Yes	No
Prepare for troubleshooting.			No. of Entries		1	2	2	3	3
			Tail / Job No.						
	k.	Gaseous systems – pneumatic	LAME Sign.						
			Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries		1	2	<u>)</u>	:	3
			Tail / Job No.						
	I.	Gaseous systems – air cycle air conditioning	LAME Sign.						
	, , , , , , , , , , , , , , , , , , ,	Date							
			Simulated	Yes	No	Yes	No	Yes	No

# Performance Criteria:

1.1 Relevant maintenance documentation and modification status, including system defect/service difficulty reports, where relevant, are interpreted to identify an unserviceability.

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UNIT MEA320: Test and tro	ublesh	noot aircraft hydro-mechanical, gaseous and landing gear systems and	components						
			No. of Entries	-	L	2	)	(1)	3
			Tail / Job No.						
	m.	Gaseous systems – pressurisation	LAME Sign.						
			Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries	-	L	2	) -	3	3
			Tail / Job No.						
	n.	Gaseous systems – Fire Extinguishing	LAME Sign.						
			Date						
1. Cont'd			Simulated	Yes	No	Yes	No	Yes	No
Prepare for troubleshooting.		Gaseous systems components – gauges (direct reading),	No. of Entries	-	L	2	<u>-</u>	3	3
	Ο.		Tail / Job No.						
		temperature sensors, pressurisation controllers and temperature	LAME Sign.						
		controllers	Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries		_	2	-	3	3
	p.	Gaseous systems components – heat exchangers, pressure vessels,	Tail / Job No.						
	, , , , , , , , , , , , , , , , , , , ,	condensers, compressors, expansion turbines, humidifiers, valves	LAME Sign.						
		and actuators	Date						
			Simulated	Yes	No	Yes	No	Yes	No

# Performance Criteria:



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UNIT MEA320: Test and tro	ubles	hoot aircraft hydro-mechanical, gaseous and landing gear systems and	components					
			No. of Entries	1		2		3
	_	Cospers systems companyed trivid and flevible pinclines and	Tail / Job No.					
	q.	Gaseous systems components – rigid and flexible pipelines and fittings	LAME Sign.					
		nttings	Date					
			Simulated	Yes No	Ye	S No	Yes	No
			No. of Entries	1		2		3
			Tail / Job No.					
	r.	Gaseous systems components – ducting	LAME Sign.					
			Date					
1. Cont'd			Simulated	Yes No	Ye	S No	Yes	No
Prepare for troubleshooting.			No. of Entries	1		2		3
			Tail / Job No.					
	s.	Mechanical operating and locking systems	LAME Sign.					
			Date					
			Simulated	Yes No	Ye	No.	Yes	No
			No. of Entries	1		2	;	3
	t.	Cables, pulleys, guides, fairleads, tension regulators, chains and	Tail / Job No.					
	ι.	sprockets	LAME Sign.					
		sprockets	Date					
			Simulated	Yes No	Ye	S No	Yes	No

# Performance Criteria:



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Name of Assessed Person: Registration:

UNIT MEA320: Test and tro	INIT MEA320: Test and troubleshoot aircraft hydro-mechanical, gaseous and landing gear systems and components													
		No. of Entries	1	2		3	3							
1. Cont'd Prepare for troubleshooting.		Tail / Job No.												
	u. Push/pull rods, torque tubes, bellcranks, screw jacks, clutches, springs, bearings and gears.	LAME Sign.												
		Date												
		Simulated	Yes No	Yes I	No	Yes	No							

# Performance Criteria:



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Name of Assessed Person: Registration:

UNIT MEA320: Test and tro	ubles	hoot aircraft hydro-mechanical, gaseous and landing gear systems and	components						
			No. of Entries	1	L	2	2	3	3
		Undraulia system budraulia accumulators filters reservoirs values	Tail / Job No.						
	a.	Hydraulic system - hydraulic accumulators, filters, reservoirs, valves, pumps, motors, actuators, regulators and direct reading gauges	LAME Sign.						
		pumps, motors, actuators, regulators and direct reading gauges	Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries	1	L	2	2	3	3
			Tail / Job No.						
	b.	Hydraulic system - rigid and flexible pipelines, hoses and fittings	LAME Sign.						
2.			Date						
Test hydro-mechanical, mechanical, gaseous and			Simulated	Yes	No	Yes	No	Yes	No
landing gear systems and			No. of Entries	1	L	2	2	3	3
components.	c.	Fuel system - filters, valves, pumps, rigid and flexible storage	Tail / Job No.						
	L.	cells/tanks	LAME Sign.						
		cens, tanks	Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries	1	L	2	<u>)</u>	3	3
			Tail / Job No.					<u> </u>	
	d.	Fuel system - rigid and flexible pipelines, hoses and fittings	LAME Sign.					<u> </u>	
		Date					<u> </u>		
			Simulated	Yes	No	Yes	No	Yes	No

- 2.1 The aircraft and hydro-mechanical, mechanical, gaseous and landing gear systems are correctly prepared in accordance with specified procedures for the application of power.
- 2.2 Power is applied, and system and components functionally tested in accordance with specified procedures for evidence of malfunction or leaks while applying all relevant work health and safety (WHS) procedures.
- 2.3 System calibration or adjustments are performed in accordance with specified procedures.



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Name of Assessed Person: Registration:

UNIT MEA320: Test and tro	oubles	hoot aircraft hydro-mechanical, gaseous and landing gear systems and	components						
			No. of Entries	1	L	2	<u>)</u>	(1)	3
			Tail / Job No.						
	e.	Landing gear systems – retraction systems *(Note 1)	LAME Sign.						
			Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries	1	L	2	2	3	3
			Tail / Job No.						
2. Cont'd	f.	Landing gear systems - steering systems *(Note 1)	LAME Sign.						
		Si	Date						
Test hydro-mechanical, mechanical, gaseous and			Simulated	Yes	No	Yes	No	Yes	No
landing gear systems and			No. of Entries	1	L	2	2	3	3
components.	۱ ,	Landing gear systems - brake systems, including anti-skid, where	Tail / Job No.						
	g.	applicable *(Note 1)	LAME Sign.						
		applicable (Note 1)	Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries	1	L	2	<u>)</u>	3	3
			Tail / Job No.						
	h.	Landing gear components – wheel assembly *(Note 1)	LAME Sign.						
		Date							
			Simulated	Yes	No	Yes	No	Yes	No

### Performance Criteria:

- 2.1 The aircraft and hydro-mechanical, mechanical, gaseous and landing gear systems are correctly prepared in accordance with specified procedures for the application of power.
- 2.2 Power is applied, and system and components functionally tested in accordance with specified procedures for evidence of malfunction or leaks while applying all relevant work health and safety (WHS) procedures.
- 2.3 System calibration or adjustments are performed in accordance with specified procedures.

\*(Note 1) Not applicable to aircraft fitted with skids or floats



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NIT MEA320: Test and tro	ubles	hoot aircraft hydro-mechanical, gaseous and landing gear systems and	components						
			No. of Entries	1		2	<u>)</u>	3	3
			Tail / Job No.						
	i.	Landing gear components – brake units *(Note 1)	LAME Sign.						
			Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries	1		2	2	3	3
		Tail / Job No.							
2. Cont'd	j.	Landing gear components – struts/oleos *(Note 1)	LAME Sign.						
	<del> </del>	Date							
Test hydro-mechanical, mechanical, gaseous and			Simulated	Yes	No	Yes	No	Yes	No
landing gear systems and			No. of Entries	1		2	2	3	3
components.			Tail / Job No.						
	k.	Gaseous systems – pneumatic	LAME Sign.						
			Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries	1		2	<u>)</u>	3	3
			Tail / Job No.						
	I.	Gaseous systems – air cycle air conditioning	LAME Sign.						
		, ,	Date						
			Simulated	Yes	No	Yes	No	Yes	No

### **Performance Criteria:**

- 2.1 The aircraft and hydro-mechanical, mechanical, gaseous and landing gear systems are correctly prepared in accordance with specified procedures for the application of power.
- 2.2 Power is applied, and system and components functionally tested in accordance with specified procedures for evidence of malfunction or leaks while applying all relevant work health and safety (WHS) procedures.
- 2.3 System calibration or adjustments are performed in accordance with specified procedures.

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			No. of Entries	1	-	2	2	3	3
			Tail / Job No.						
	m.	Gaseous systems – pressurisation	LAME Sign.						
			Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries	1	-	2	2	(1)	3
			Tail / Job No.						
	n.	Gaseous systems – Fire Extinguishing	LAME Sign.						
2. Cont'd			Date						
Test hydro-mechanical,			Simulated	Yes	No	Yes	No	Yes	No
mechanical, gaseous and landing gear systems and			No. of Entries	1	-	2	2	(1)	3
components.	o.	Gaseous systems components – gauges (direct reading),	Tail / Job No.						
		temperature sensors, pressurisation controllers and temperature	LAME Sign.						
		controllers	Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries	1	-	2	2	3	3
	p.	Gaseous systems components – heat exchangers, pressure vessels,	Tail / Job No.						
		LAME Sign.							
		Date							
			Simulated	Yes	No	Yes	No	Yes	No

- 2.1 The aircraft and hydro-mechanical, mechanical, gaseous and landing gear systems are correctly prepared in accordance with specified procedures for the application of power.
- 2.2 Power is applied, and system and components functionally tested in accordance with specified procedures for evidence of malfunction or leaks while applying all relevant work health and safety (WHS) procedures.
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			No. of Entries	1		2	2	:	3
		Consequence of the control of the co	Tail / Job No.						
	q.	Gaseous systems components – rigid and flexible pipelines and	LAME Sign.						
		fittings	Date						
			Simulated	Yes N	lo	Yes	No	Yes	No
			No. of Entries	1		2	<u>)</u>	:	3
	r. Gaseous systems components – ducting	Tail / Job No.							
2. Cont'd		LAME Sign.							
		Date							
Test hydro-mechanical,			Simulated	Yes N	lo	Yes	No	Yes	No
mechanical, gaseous and landing gear systems and		No. of Entries	1		2	<u>)</u>	:	3	
components.			Tail / Job No.						
compensation.	s.	Mechanical operating and locking systems	LAME Sign.						
			Date						
			Simulated	Yes N	lo	Yes	No	Yes	No
			No. of Entries	1		2	2	:	3
		Cables nullays guides fairleads tension regulators chains and	Tail / Job No.						
	t.	Cables, pulleys, guides, fairleads, tension regulators, chains and	LAME Sign.						
	sprockets	Date							
			Simulated	Yes N	lo	Yes	No	Yes	No

- 2.1 The aircraft and hydro-mechanical, mechanical, gaseous and landing gear systems are correctly prepared in accordance with specified procedures for the application of power.
- 2.2 Power is applied, and system and components functionally tested in accordance with specified procedures for evidence of malfunction or leaks while applying all relevant work health and safety (WHS) procedures.
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UNIT MEA320: Test and troubleshoot aircraft hydro-mechanical, gaseous and landing gear systems and components													
2. Cont'd			No. of Entries	1		2		(1)	3				
Test hydro-mechanical,		Duck / will useds to work to be a ballowed to some visit show about the	Tail / Job No.										
mechanical, gaseous and	u. Push/pull rods, torque tubes, bellcranks, screw jacks, clutches,	LAME Sign.											
landing gear systems and		springs, bearings and gears.	Date										
components.			Simulated	Yes	No	Yes	No	Yes	No				

- 2.1 The aircraft and hydro-mechanical, mechanical, gaseous and landing gear systems are correctly prepared in accordance with specified procedures for the application of power.
- 2.2 Power is applied, and system and components functionally tested in accordance with specified procedures for evidence of malfunction or leaks while applying all relevant work health and safety (WHS) procedures.
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UNIT MEA320: Test and tro	oubles	hoot aircraft hydro-mechanical, gaseous and landing gear systems and	components						
			No. of Entries	1		2	2	(1)	3
		Undurantia sustana handusantia sasannantatana filtana masamasina walana	Tail / Job No.						
	a.	Hydraulic system - hydraulic accumulators, filters, reservoirs, valves, pumps, motors, actuators, regulators and direct reading gauges	LAME Sign.						
		pumps, motors, actuators, regulators and direct reading gauges	Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries	1		2	2	(1)	3
3. Troubleshoot hydro- mechanical, mechanical, gaseous and landing gear systems and components			Tail / Job No.						
	b.	Hydraulic system - rigid and flexible pipelines, hoses and fittings	LAME Sign.						
			Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries	1		2	2	```	3
	c.	Fuel system - filters, valves, pumps, rigid and flexible storage cells/tanks	Tail / Job No.						
			LAME Sign.						
		Cells/ taliks	Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries	1	L	2	2		3
			Tail / Job No.						
	d.	Fuel system - rigid and flexible pipelines, hoses and fittings	LAME Sign.						
			Date						
			Simulated	Yes	No	Yes	No	Yes	No

- 3.1 Available information from maintenance documentation and inspection and test results is used, where necessary, to assist in fault determination.
- 3.2 Maintenance manual fault diagnosis guide and logical processes are used to ensure efficient and accurate troubleshooting to line replacement level.
- 3.3 Specialist advice is obtained, where required, to assist with the troubleshooting process.
- 3.4 Hydro-mechanical, mechanical, gaseous and landing gear system and component faults are located, and the causes of the faults are clearly identified and correctly recorded in maintenance documentation, where required.
- 3.5 Fault rectification requirements are determined to assist in planning the repair or adjustment.



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UNIT MEA320: Test and troubleshoot aircraft hydro-mechanical, gaseous and landing gear systems and components										
			No. of Entries	1		2	<u>)</u>		3	
			Tail / Job No.							
	e.	Landing gear systems – retraction systems *(Note 1)	LAME Sign.							
			Date							
			Simulated	Yes 1	No	Yes	No	Yes	No	
3. Cont'd Troubleshoot hydro-			No. of Entries	1		2	2		3	
			Tail / Job No.							
	f.	f. Landing gear systems - steering systems *(Note 1)	LAME Sign.							
			Date							
			Simulated	Yes 1	No	Yes	No	Yes	No	
mechanical, mechanical, gaseous and landing gear			No. of Entries	1		2	2		3	
systems and components	g.	g. Landing gear systems - brake systems, including anti-skid, where applicable *(Note 1)	Tail / Job No.							
systems and components			LAME Sign.							
			Date							
			Simulated	Yes 1	No	Yes	No	Yes	No	
			No. of Entries	1		2	<u>)</u>	3		
			Tail / Job No.							
	h.	Landing gear components – wheel assembly *(Note 1)	LAME Sign.							
			Date							
		Simulated	Yes 1	N٥	Yes	No	Yes	No		

### **Performance Criteria:**

- 3.1 Available information from maintenance documentation and inspection and test results is used, where necessary, to assist in fault determination.
- 3.2 Maintenance manual fault diagnosis guide and logical processes are used to ensure efficient and accurate troubleshooting to line replacement level.
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UNIT MEA320: Test and tr	oubles	shoot aircraft hydro-mechanical, gaseous and landing gear systems an	d components						
			No. of Entries	1	L	2	2		3
			Tail / Job No.						
	i.	Landing gear components – brake units *(Note 1)	LAME Sign.						
			Date						
			Simulated	Yes	No	Yes	No	Yes	No
3. Cont'd Troubleshoot hydro- mechanical, mechanical, gaseous and landing gear			No. of Entries	1	L	2	2	3	3
			Tail / Job No.						
	j.	j. Landing gear components – struts/oleos *(Note 1)	LAME Sign.						
			Date						
			Simulated	Yes	No	Yes	No	Yes	No
		k. Gaseous systems – pneumatic	No. of Entries	1	L	1	2	3	3
systems and components			Tail / Job No.						
	k.		LAME Sign.						
			Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries	1	L	2		3	
			Tail / Job No.						
	I.	Gaseous systems – air cycle air conditioning	LAME Sign.						
			Date						
			Simulated	Yes	No	Yes	No	Yes	No

#### Performance Criteria:

- 3.1 Available information from maintenance documentation and inspection and test results is used, where necessary, to assist in fault determination.
- 3.2 Maintenance manual fault diagnosis guide and logical processes are used to ensure efficient and accurate troubleshooting to line replacement level.
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UNIT MEA320: Test and troubleshoot aircraft hydro-mechanical, gaseous and landing gear systems and components									
			No. of Entries	1		2		3	
			Tail / Job No.						
	m.	Gaseous systems – pressurisation	LAME Sign.						
			Date						
			Simulated	Yes No	Yes	No	Yes	No	
3. Cont'd Troubleshoot hydro- mechanical, mechanical, gaseous and landing gear			No. of Entries	1		2	:	3	
			Tail / Job No.						
	n.	Gaseous systems – Fire Extinguishing	LAME Sign.						
			Date						
			Simulated	Yes No	Yes	No	Yes	No	
			No. of Entries	1		2	;	3	
systems and components	0.	Gaseous systems components – gauges (direct reading),	Tail / Job No.						
,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,		temperature sensors, pressurisation controllers and temperature controllers	LAME Sign.						
			Date						
			Simulated	Yes No	Yes	No	Yes	No	
			No. of Entries	1		2	3		
	p.	Gaseous systems components – heat exchangers, pressure vessels,	Tail / Job No.						
		condensers, compressors, expansion turbines, humidifiers, valves	LAME Sign.						
		and actuators							
			Simulated	Yes No	Yes	No	Yes	No	

#### Performance Criteria:

- 3.1 Available information from maintenance documentation and inspection and test results is used, where necessary, to assist in fault determination.
- 3.2 Maintenance manual fault diagnosis guide and logical processes are used to ensure efficient and accurate troubleshooting to line replacement level.
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			No. of Entries	1	L	2	2	(1)	3
	_		Tail / Job No.						
	q.	Gaseous systems components – rigid and flexible pipelines and fittings	LAME Sign.						
		nttings	Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries	1	L	2	2	3	3
3. Cont'd Troubleshoot hydro- mechanical, mechanical, gaseous and landing gear		r. Gaseous systems components – ducting	Tail / Job No.						
	r.		LAME Sign.						
			Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries	1	L	2	2	3	3
systems and components	S.	s. Mechanical operating and locking systems	Tail / Job No.						
o, o o o o o o o o o o o o o o o o o o			LAME Sign.						
			Date						
			Simulated	Yes	No	Yes	No	Yes	No
			No. of Entries	1	L	2	2 3		3
	_	Cables mullous guides fairleads tansian regulators shains and	Tail / Job No.						
	t.	Cables, pulleys, guides, fairleads, tension regulators, chains and sprockets	LAME Sign.						
		Sprockets	Date						
			Simulated	Yes	No	Yes	No	Yes	No

#### **Performance Criteria:**

- 3.1 Available information from maintenance documentation and inspection and test results is used, where necessary, to assist in fault determination.
- 3.2 Maintenance manual fault diagnosis guide and logical processes are used to ensure efficient and accurate troubleshooting to line replacement level.
- 3.3 Specialist advice is obtained, where required, to assist with the troubleshooting process.
- 3.4 Hydro-mechanical, mechanical, gaseous and landing gear system and component faults are located, and the causes of the faults are clearly identified and correctly recorded in maintenance documentation, where required.
- 3.5 Fault rectification requirements are determined to assist in planning the repair or adjustment.

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Name of Assessed Person: Registration:

UNIT MEA320: Test and troubleshoot aircraft hydro-mechanical, gaseous and landing gear systems and components									
3. Cont'd			No. of Entries	1		2	2	3	
Troubleshoot hydro-		Duck / ault vada tavaua tukaa hallavanka aavau jaaka akitakaa	Tail / Job No.						
mechanical, mechanical,	u.	springs, bearings and gears.	LAME Sign.						
gaseous and landing gear			Date						
systems and components			Simulated	Yes	No	Yes	No	Yes	No

- 3.1 Available information from maintenance documentation and inspection and test results is used, where necessary, to assist in fault determination.
- 3.2 Maintenance manual fault diagnosis guide and logical processes are used to ensure efficient and accurate troubleshooting to line replacement level.
- 3.3 Specialist advice is obtained, where required, to assist with the troubleshooting process.
- 3.4 Hydro-mechanical, mechanical, gaseous and landing gear system and component faults are located, and the causes of the faults are clearly identified and correctly recorded in maintenance documentation, where required.
- 3.5 Fault rectification requirements are determined to assist in planning the repair or adjustment.



AA TT PRO 01a

Name of Assessed Person: Registration:

Certification of Underpinning Knowledge and Skills to Test and troubleshoot aircraft hydro-mechanical, gaseous and landing gear systems and components

A person cannot be assessed as competent until it can be demonstrated to the satisfaction of the workplace assessor that the relevant elements of this unit of competency are being achieved under routine supervision on each type of system and on at least one (1) component of each group listed in the assessment conditions a) to u) that are applicable to the enterprise. This shall be established via the records in the Log of Industrial Experience and Achievement or, where appropriate, an equivalent Industry Evidence Guide (for details refer to the Companion Volume Implementation Guide).

UNIT MEA320: Test and troubleshoot aircraft hydro-mechanical, gaseous and landing gear systems and components	Date/ MTO Stamp
Evidence has been confirmed of the attainment of the following pre-requisite units of competency (as they are relate to attainment of the elements of competency specified in this unit).	d
318	
Evidence has been confirmed of the knowledge requirements for this unit as delivered by a CASR 147 Approved Organisation.	
OR	
Assessment has been conducted to determine that the underpinning knowledge and skills have been achieved in accordance with the Competency Unit.	

# **Certification of Unit Completion**

I certify that I have reviewed the certification of the elements for this competency unit and that all of the competency unit requirements have been met.

Signed:	Assessor No.	MTO:	Date:



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